

## Nano-scale sensors for wall-shear-stress measurements

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Wall-bounded turbulent flows typically contain a wide range of spatio-temporal scales. Experimental measurements that fully resolve all the scales are of great value in understanding and modeling the underlying flow physics. In a boundary layer flow, wall-shear stress is a physical quantity of fundamental importance, which is not amenable to easy measurements. The present effort is aimed at the development of a novel nano-scale sensor for time-resolved wall-shear-stress measurements in a turbulent boundary layer flow. The sensor is based on the principles of thermal anemometry.<sup>1</sup> These sensors are designed and fabricated in-house using standard nano-fabrication techniques<sup>2</sup> and tested in a low-speed turbulent boundary layer flow. The sensing element is a thin platinum wire of dimension  $45\ \mu\text{m} \times 1.5\ \mu\text{m} \times 130\ \text{nm}$ , with electrical contact pads of dimension  $100\ \mu\text{m} \times 100\ \mu\text{m}$  deposited on a  $\text{SiO}_2/\text{Si}$  substrate. Figure 1(a) shows a scanning electron microscope image (SEM) of the sensor with two independent filaments. The sensor is placed flush with the wall (see figure 1(b)), and is electrically heated (similar to a hot wire probe). Convective transfer of the ohmic heat to the flow informs the local wall-shear-stress condition. Quantitative inference of wall-shear stress requires a calibration exercise, for which a standard acoustic plane-wave tube is employed.<sup>3</sup> Details of the sensor design and fabrication, along with experimental results from a  $Re_\tau = 800$  (friction Reynolds number) boundary layer flow will be presented at the conference.

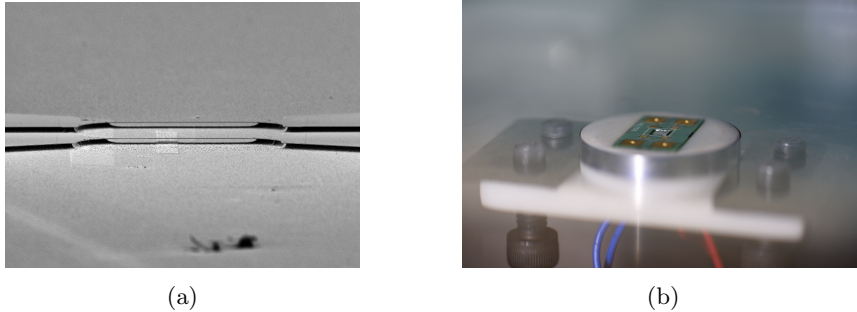


Figure 1: (a) A representative SEM image of the sensor; (b) Packaged sensor integrated on a flat plate wind tunnel model for boundary layer studies.

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<sup>1</sup>H. H. Bruun, *Hot-wire Anemometry*, Oxford University Press (1995)

<sup>2</sup>Vallikivi and Smits, *IEEE Journal of Microelectromechanical Systems*, vol. 23, no. 4 (2014)

<sup>3</sup>Sheplak et al., *AIAA Journal*, vol. 39, no. 5 (2001)